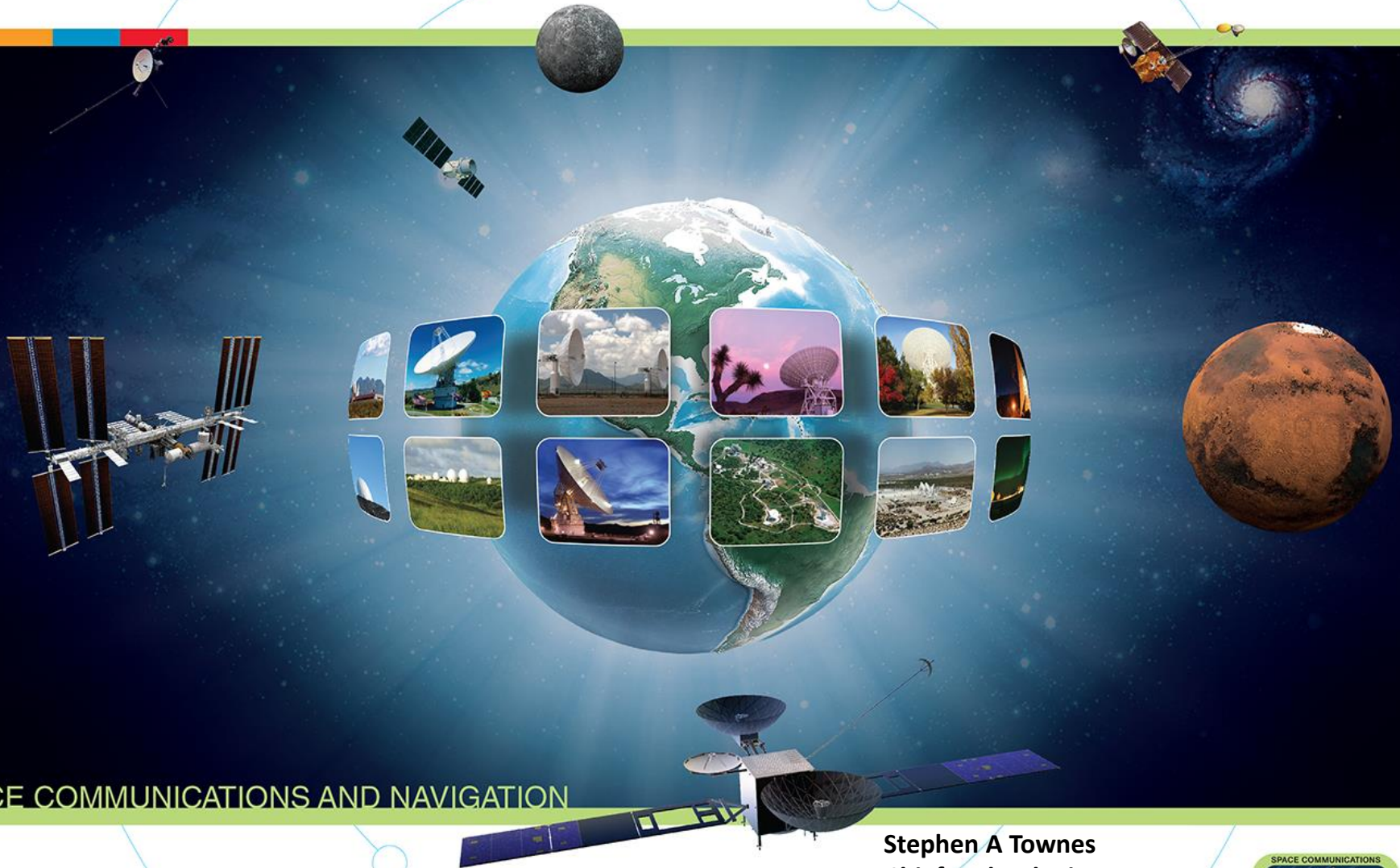


# ESA-NASA/JPL Technical Interchange

National Aeronautics and  
Space Administration



SPACE COMMUNICATIONS AND NAVIGATION

Stephen A Townes  
Chief Technologist  
Interplanetary Network Directorate  
Jet Propulsion Laboratory  
California Institute of Technology





National Aeronautics and  
Space Administration



Jet Propulsion Laboratory  
California Institute of Technology

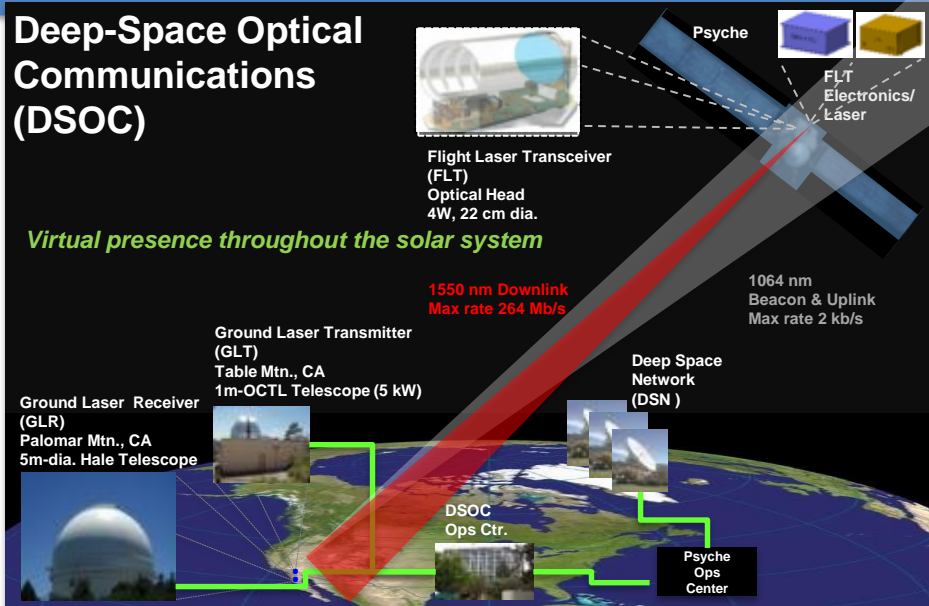
# Deep Space Optical Communications (DSOC)

Bill Klipstein, Project Manager  
Abi Biswas, Project Technologist



# Deep Space Optical Communications

## Deep-Space Optical Communications (DSOC)



## Objectives:

- **Demonstrate deep space optical communication capability**
  - Designed for 0.1 to 2.7 AU
  - Sun-Earth Probe Angle > 25° (TBC)
- **Develop a Flight Laser Transceiver (FLT) for accommodation on Psyche spacecraft**
  - Downlink data-rate of **0.256 - 200 Mb/s**
  - Uplink data-rate of **[2 kb/s]**
  - Prime demonstration duration **2 years**
- **Develop ground network**
  - GLT for transmitting laser beacon out to 2.7 AU
  - GLR retrofitted with photon counting receiver
  - Mission Operations System

**Project Manager (PM) :** Bill Klipstein, (818)-354-2245

**Project Technologist (PT):** Abi Biswas

## Sponsors:

STMD/TDM (flight), HEOMD/SCaN (ground), SMD (host)

## Facilities:

Optical Comm and Environmental Test Labs at JPL

Vendor site Labs and test facilities

Optical Communication Telescope Laboratory (OCTL)

Caltech Optical Observatories/Hale Telescope Observatory

Psyche mission host

## Key Milestones:

**FY14-16** GCD Technology Development Phase

**FY17** Phase A Start, SRR/MDR

**FY18** PDR

**FY19** CDR

**FY20** Downlink I+T start at JPL

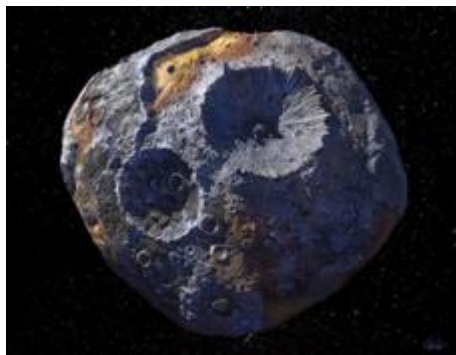
**FY21** start I+T at Hale and OCTL

**FY22** ORR

**FY23-25** Launch, Ops

# DSOC on Psyche

- Explores a metal world: a nickel-iron protoplanet core, a Trojan asteroid at  $\sim 3.3$  AU
- PI: Lindy Elkins-Tanton, ASU
- PM: Henry Stone, JPL
- S/C: Space-Systems Loral
- Excellent opportunity for a robust demonstration of optical comm out to  $\sim 3$  AU in 1<sup>st</sup> 2 years
- Psyche's schedule:
  - Launch in Oct 2023 corresponds to delivery of DSOC to S/C in June, 2022



<https://sese.asu.edu/research/psyche>

<https://www.facebook.com/Psyche-Mission-1598743977091187/>

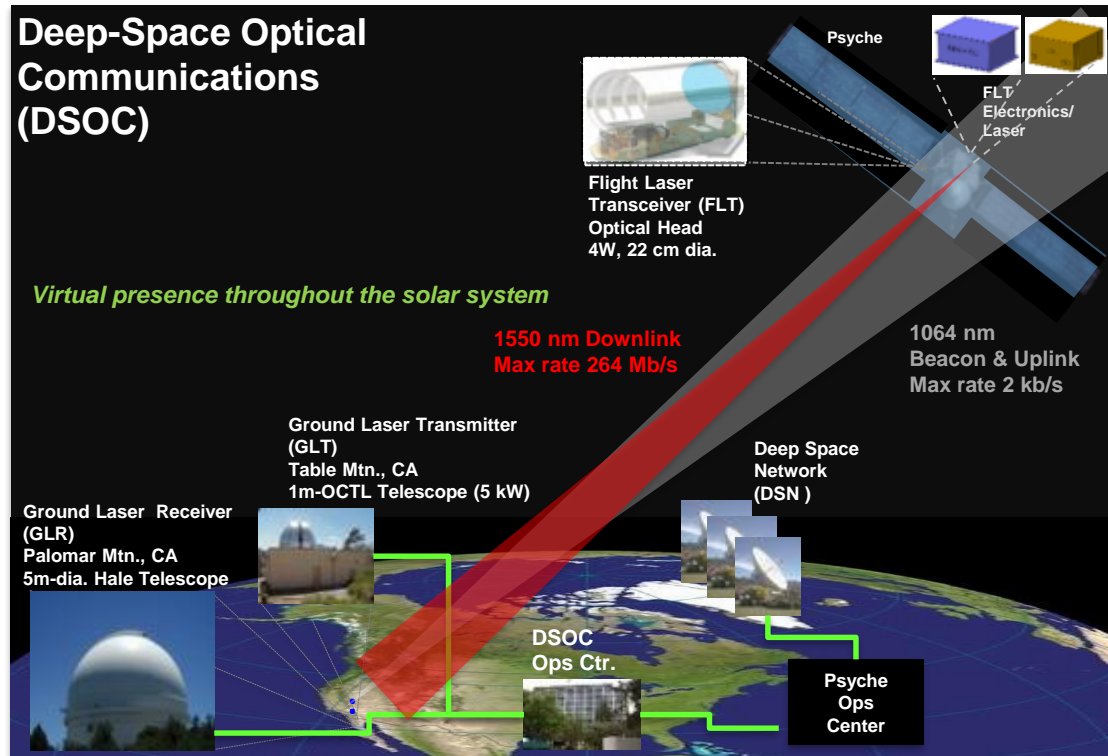


- DSOC to operate during cruise
- Earth flyby at 1 year followed by Mars flyby at 1.75 years is near ideal for a 2 year DSOC demo
  - System shake out and demonstration of high data rates at closer ranges
  - Reaches farthest design distance inside 2 year mission lifetime



# Overview

- **Beacon based architecture with existing ground assets for cost-effective demo**
  - Beacon laser beam serves as
    - A pointing reference
    - Low-rate uplink data carrier
    - Requires sensitive detector for dim beacon from deep-space range
  - Downlink is serially concatenated pulse position modulation (SCPPM)
    - Received with a photon-counting detector array retrofitted to 5 m diameter Hale telescope
- **Link demonstration constraints**
  - Ground Laser Receiver restricted to 25°(TBC) SEP angle
    - Expect link outages
- **Psyche Mission to host DSOC**
  - Psyche
  - Ranges from ~ 0.1 - 3 AU
  - Diverse sun angles, air-mass, time of day/year

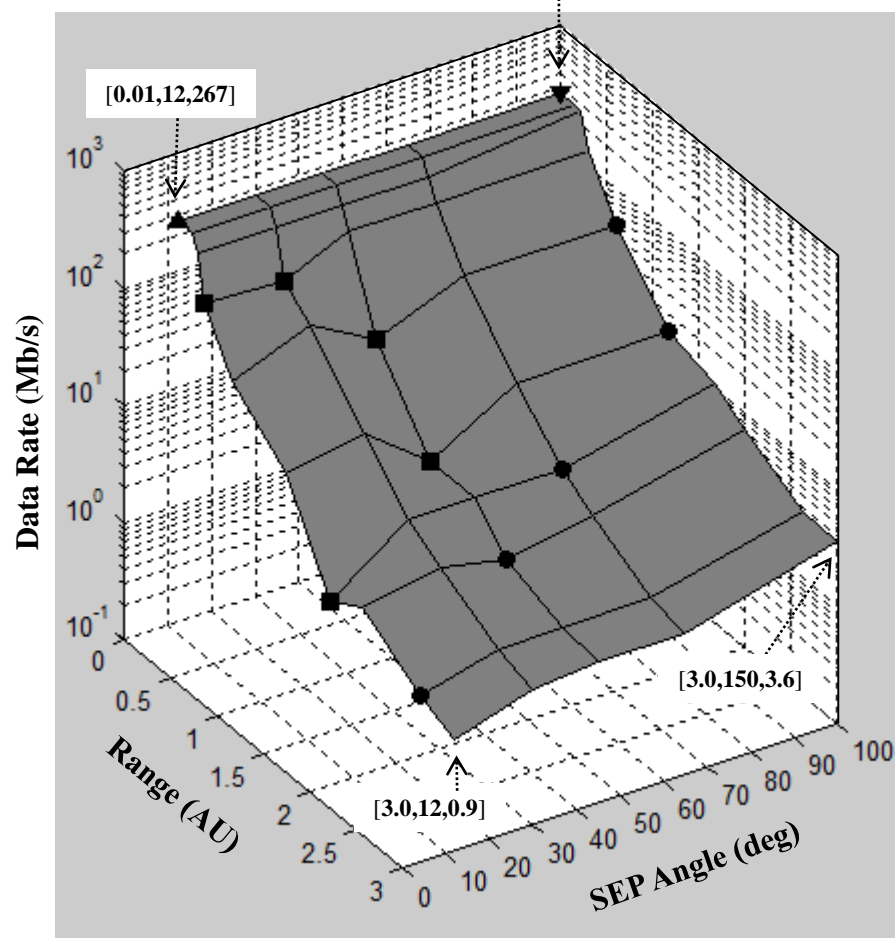


# Notional Link Performance

[0.01,150,267]

- Range and additive background dependency photon-counting link throughput

- 1550 nm
- 4 W laser average power
- 22-cm aperture diameter
- Serially concatenated pulse position modulation (SCPPM)
- Nominal atmospheric conditions
- 60° zenith angle
- $r_0$  (day) = 3.6 cm
- $r_0$  (night) = 5 cm
- Min Sep 12°
- 5 m dia. ground collector
- Photon-counting receiver
- At all other ranges 3-4.5 dB margin



- ▲ Earth-Sun Lagrange (L1)
- ▼ Earth-Sun Lagrange (L2)
- Venus Orbiter
- Mars Orbiter

# DSOC Space Flight Hardware

## Telescope & Optics

### Flight Laser Transceiver (FLT) Assembly

- Uplink receiver
- Downlink transmitter

### Point-Ahead Mirror

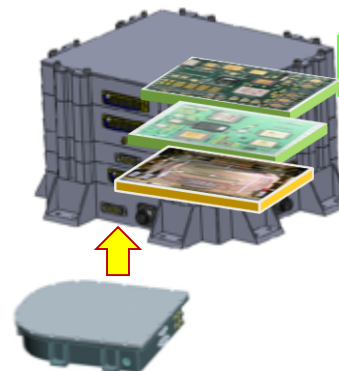


### Photon-Counting Camera



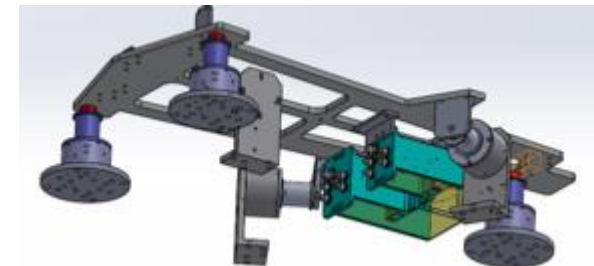
### Electronics Box

- Electronics processing & control cards, firmware, software, clock



### Laser Transmitter

- High Peak-to-average power

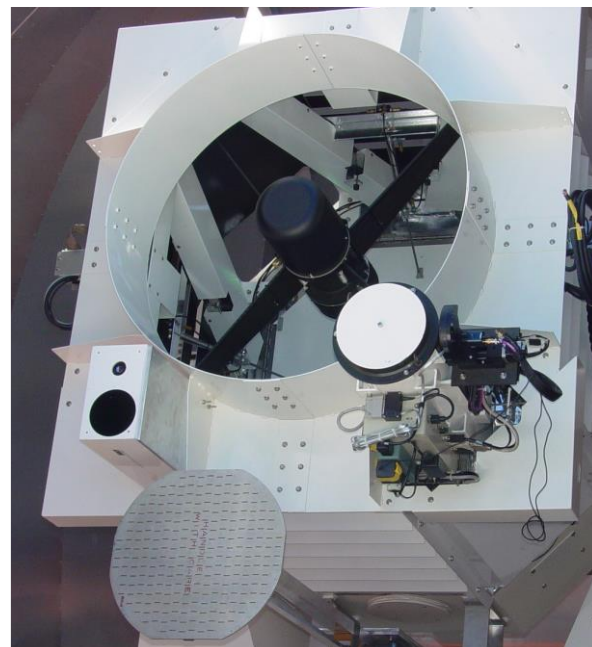
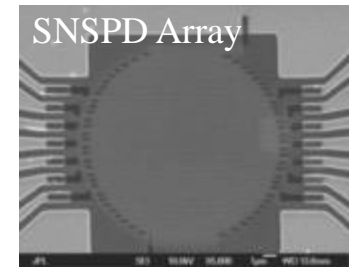


- ### Isolation Pointing Assembly (IPA)
- Steering and line-of-sight (LOS) stabilization

- Deep-space optical communications Flight Laser Transceiver (FLT) characteristics
  - *Photon-efficient communications*
  - *Off-axis Gregorian telescope*
  - *Isolation Pointing Assembly*
  - *Photon Counting camera*
  - *High peak-to-average power laser transmitter*
  - *Electronics*
- Assembly level technologies
  - *Developed by vendors and at JPL*
- Integration & testing at JPL

# DSOC Ground Characteristics

- Existing telescopes to be used for transmitter (OCTL) and Receiver (Hale telescope at Palomar Mountain)
- Laser transmitter for OCTL - baseline wavelength is 1064 nm
- Photon Counting receiver - Tungsten silicide (WSi) superconducting nanowire single photon detectors (SNSPD)
  - Developed at JPL
  - Large effective diameter array with read-out integration circuit



Optical Communication  
Telescope Laboratory  
(OCTL)  
1 meter diameter

Palomar 5 meter Telescope

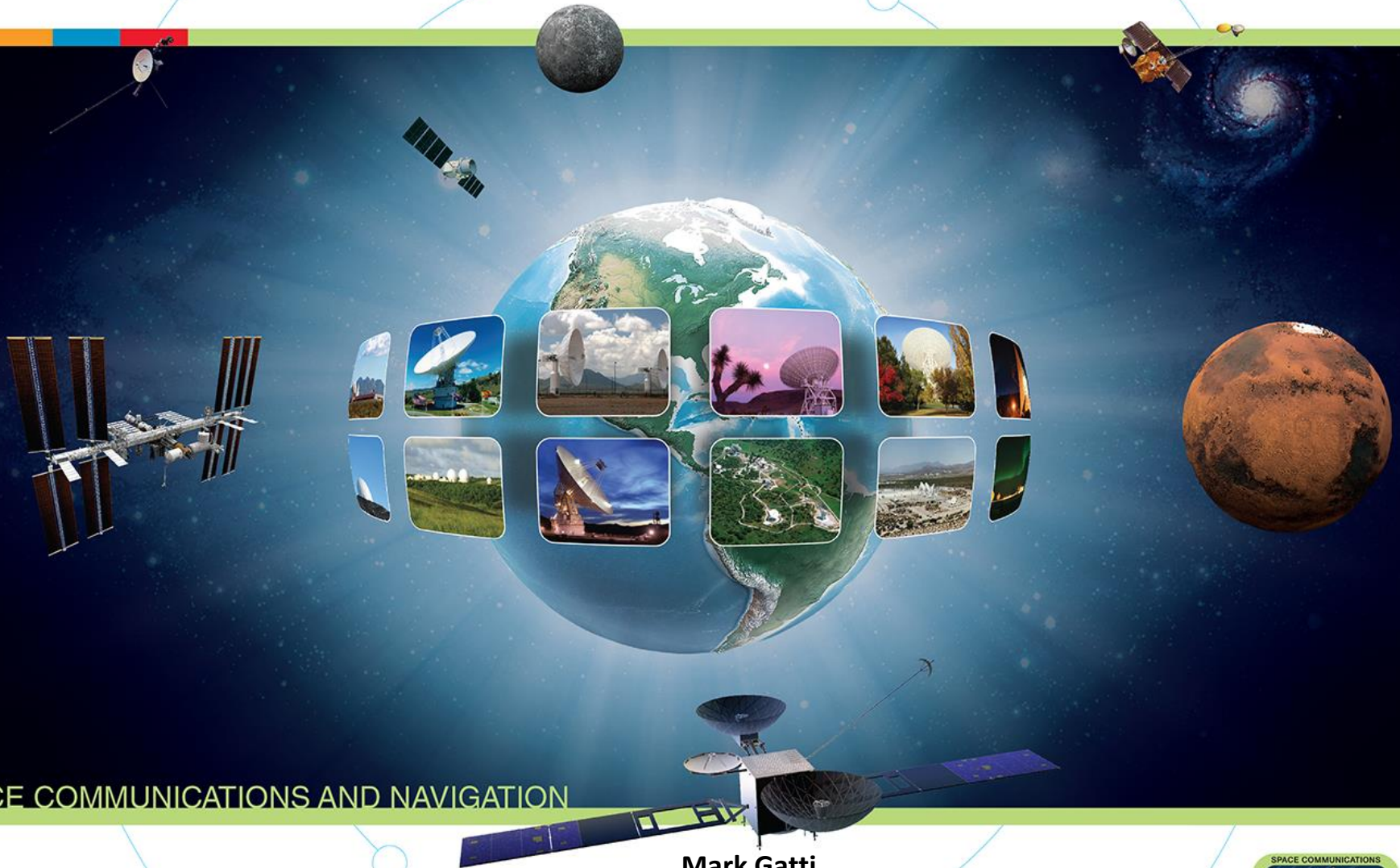




# Summary

- **Technology Demonstration (TD) opportunity for laser communications**
  - Psyche mission to host TD
  - The DSOC Project is developing a flight and ground sub-system
  - Several new technologies will be tested and verified:
    - Space Photon Counting camera
    - Isolation Pointing Assembly
    - Flight Laser transmitter
    - Ground photon counting detector array
  - Gain deep space optical communications operational experience
    - Planning and scheduling link operations
    - Verifying link acquisition and tracking
    - Evaluating link performance under diverse conditions

# Status: Potential 34 Meter Hybrid RF-Optical Antenna



SPACE COMMUNICATIONS AND NAVIGATION

**Mark Gatti**

Manager, Ground Communications Section  
Jet Propulsion Laboratory



# Introduction



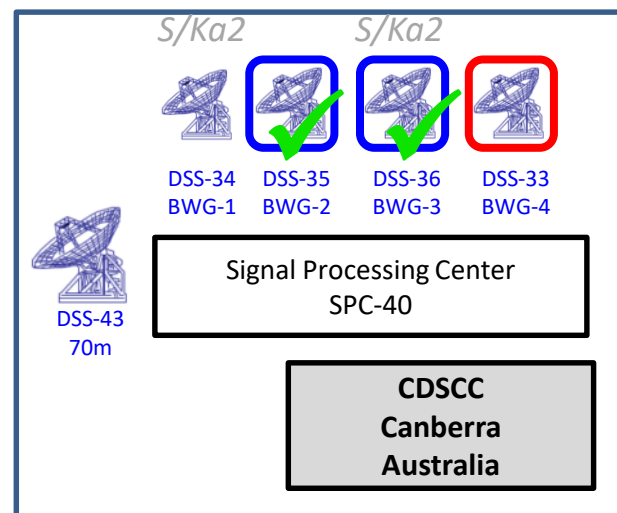
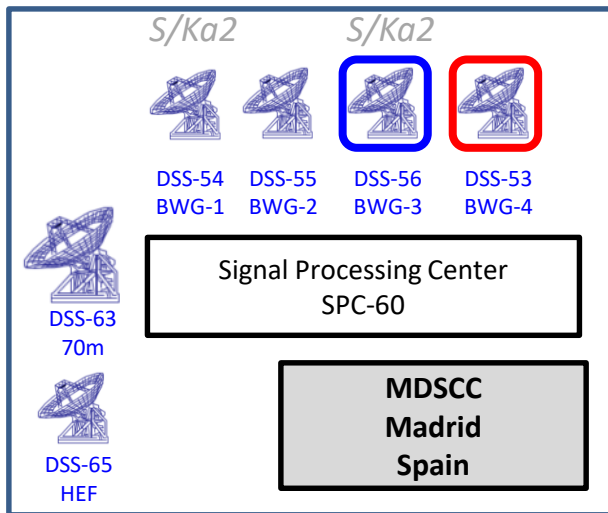
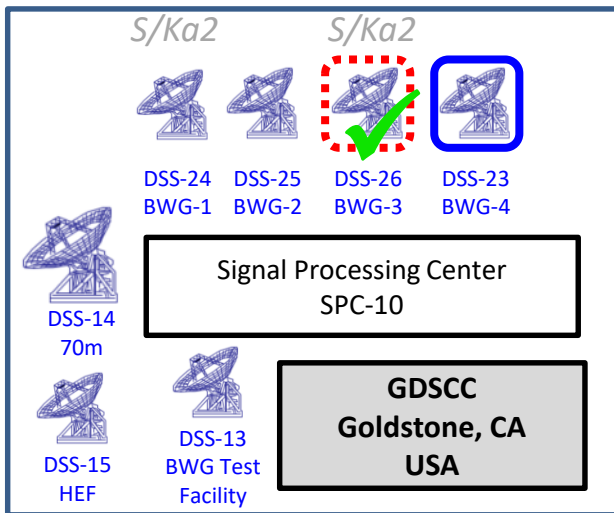
- JPL is proposing to incorporate an approximately 8 meter optical aperture into a 34 meter RF antenna
- Potentially part of the DSN Aperture Enhancement Project (DAEP)
  - No commitment from SCaN
  - All dates shown are notional but will be coordinated with DAEP
- Still in the initial systems engineering, technology development and validation stage

Thanks to Dan Hoppe and Brad Arnold for their contributions to this briefing





# DAEP Rollout



## DAEP Developments



*S* S-Band Up/Dn  
*Ka2* 26-GHz Dn  
Complete

| Station       | XX/Ka          | S              | Ka2            |
|---------------|----------------|----------------|----------------|
| DSS-26        |                | 10/2017        | (TBD)          |
| DSS-35        | 10/2014        | -              | -              |
| DSS-36        | 10/2016        | 10/2016        | (TBD)          |
| <b>DSS-56</b> | <b>03/2020</b> | <b>03/2020</b> | <b>03/2020</b> |
| DSS-53        | 10/2020        | -              | -              |
| DSS-23        | 10/2022        | -              | -              |
| DSS-33        | 10/2024        | -              | -              |



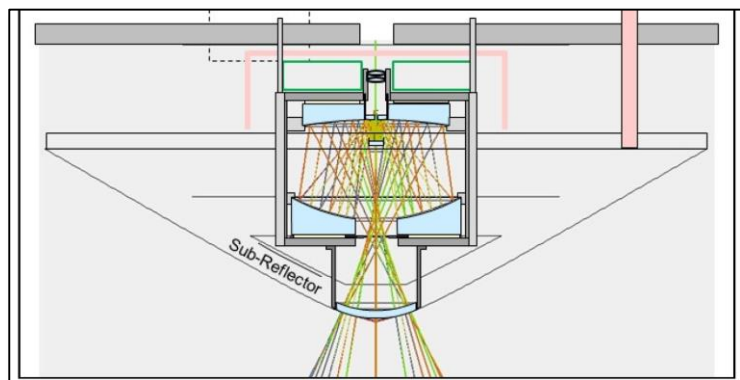
# Overview



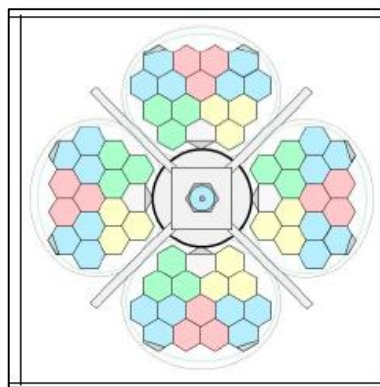
- Transformational development for SCaN for next generation deep space telecom, with feed forward for human exploration support
  - Operational Deep Space Optical capability
  - Ready to augment Discovery optical comm flight demo and early crewed missions
- Leverage earlier NASA-funded studies and tests, include key development gates minimize risk to NASA
- Implementation approach is to integrate new 8-m optical apertures with ongoing new 34m BWG antennas
- Minimize impact to baseline DAEP RF antenna schedule
  - Potentially, both hemispheres covered with optical deployments at GDSCC and CDSCC

- Optical Design

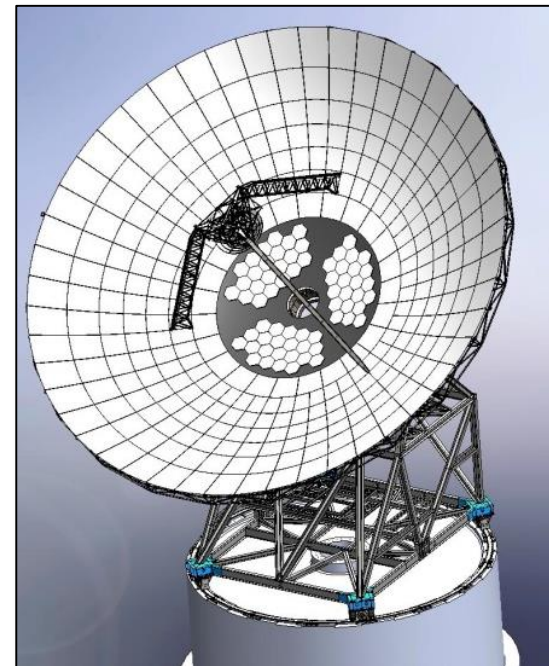
- Adapt two planned 34 m BWG for DAEP
  - Small loss to RF performance – studies ongoing
- Primary spherical mirrors replace inner RF panels
  - Small loss to RF performance – studies ongoing
- Spherical aberration correction optics and receiver package located behind RF subreflector
- Initial risk reduction development at GDSCC's DSS-13
  - Tripod vs. quadripod at production BWG



Spherical Aberration Correction Optics  
behind Subreflector



DAEP 34m Design

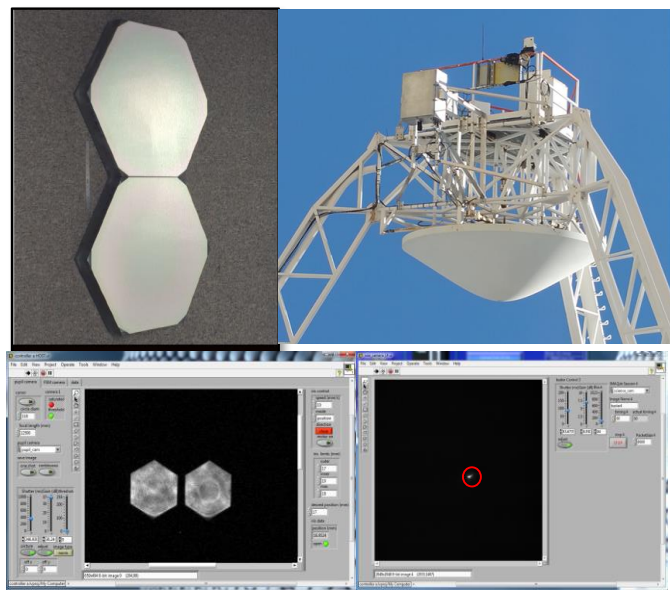


DSS-13 8 m Primary  
Surface Plan



# Developments To Date

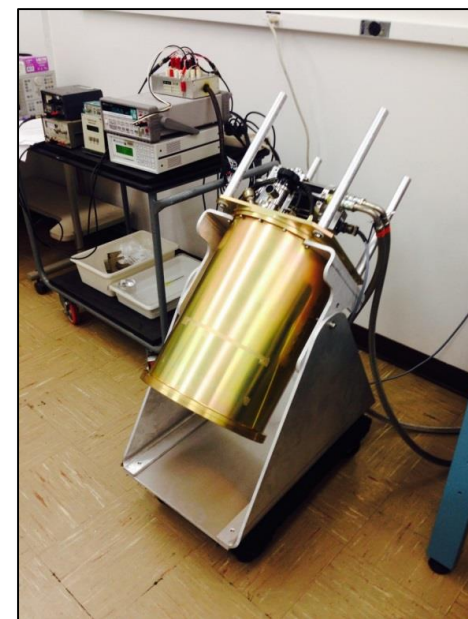
- Early optical studies with pair of 35 cm panels DSS-13 complete, fielded second generation focal plane assembly last FY
- Low temperature cryo demonstrated (0.5 K at detector)
- Completed tipping and mechanical vibration tests, vetting design



DSS-13 2-Mirror Test

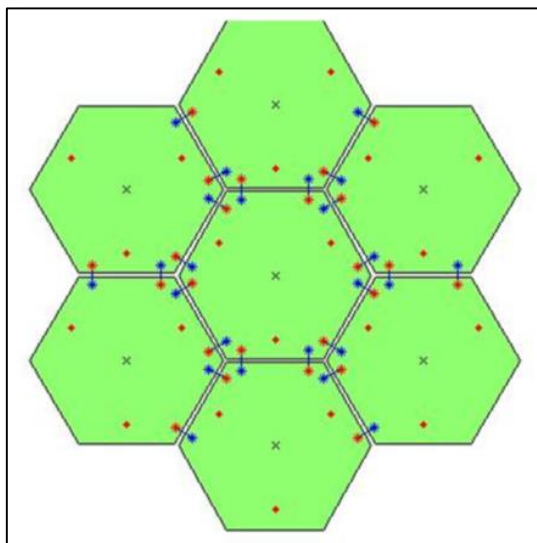


Cryo Package Prototype

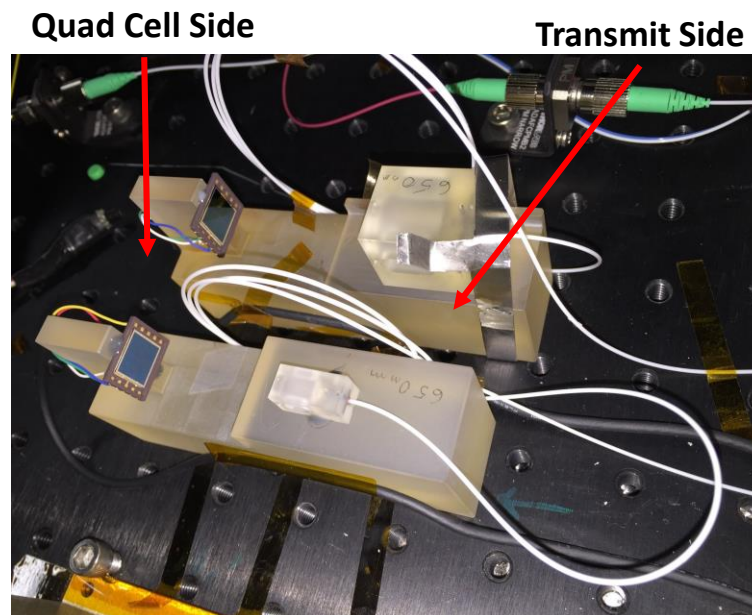


Cryo Tipping Test

- Edge sensors and control will be required to maintain overall surface alignments.
- Early development and testing of sensor system including outdoor testing is ongoing.



Edge Sensor Layout in Lattice



Early Testing of 2 Versions of the Detector System



# Proposed Development Plan



- Preliminary development & analysis FY'17 – FY'19
  - Optical system analysis
    - Stray light analysis (complete by end of April)
    - Link budgets
    - Supportable solar angles
  - CDSCC atmospheric study
    - Equipment installation plus 2-3 years data collection
  - O&M considerations
  - Refine costing estimates





# Proposed Development Plan (cont'd)



- Main technology demonstration and development gates
  - By Jan-2020
    - Seven 0.5m-element Non-Recurring Engineering study at DSS-13
    - Equivalent ~1.2 m demonstrated
    - Risk retired: Proof of concept demonstration showing alignment functionality of array with lower cost smaller optical elements
  - By Sep-2021
    - Sixteen 1.1 m production element “pod” + aberration correction optics at DSS-13
    - Equivalent ~4.15 m demonstrated
    - Risks retired: Performance of sub-array with full sized elements and optics demonstrated
  - Challenge: space based signals to test against (LCRD, stars, LEOs w/LRA, etc.)
    - Technical options will be considered
    - Psyche launches late 2022 or 2023 - augmentation to end of DSOC demonstration feasible



# Proposed Development Plan (cont'd)



- Final build out and test
  - By Oct-2023
    - 64 elements (1.1 m) installed at DSS-13
    - Optical detector and receiver build, test and installation
    - Full equivalent 8 m demonstrated
  - By Oct-2024
    - Move equipment to DSS-23
    - Full integration and test complete
    - DSS-23 RF capability transferred to operations.
    - Optical capability will initially be run as a demonstration, until experience is gained in operating and maintaining the capability



# Development Plan (cont'd)



- DSS-33
  - Start antenna build in FY'21
  - Complete full I&T and delivery to DSN by Oct-2026
- Additional requirements to modify BWG antenna
  - Preliminary analysis shows low risk forward, detailed analysis to go
  - Inner rings of RF panels not populated
  - RF subreflector reshaped, RF panels re-positioned
  - Antenna subreflector modified to accept Spherical Aberration Corrector, receiver and cryo equipment behind subreflector





# 10 m RF-Optical Hybrid



- 10-m equivalent diameter optical
  - Under study now
  - Early assessment suggests some redesign of 34m BWG would likely be required (f/D optics)



# Conclusion



- The RF/Optical DSN antenna will be a transformational development
- Clear path to implementation with feed forward for next generation deep space telecom, including human exploration
- Plan minimizes impact to RF antenna development
- Delivers DSS-23/33 8 m optical hybrid antennas in 2024 and 2026
- Initial development and plan have been prepared
- Minimizes risk to NASA resources by scheduling two key development gates